

Getting Started with LS-DYNA Implicit

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- The purpose of this tutorial is to introduce new users to the basics of LS-DYNA Implicit and to convert an existing explicit model to an implicit one.
- For demonstration purposes, the Oasys pre- and post-processing software (PRIMER, D3PLOT and T/HIS) are used although any software that supports LS-DYNA can be used (i.e. LS-PrePost).

Steps covered:

- Review LS-DYNA explicit model and results of cantilever beam, which is fixed at one end and subject to point load at free end.
- Submit eigenvalue analysis to check and eliminate rigid body modes.
- Submit linear static analysis.
- Verify results against hand calculations.

An incomplete input deck is provided: **linear_static_explicit.key**

Pre-Processing: PRIMER Environment

LS-DYNA Implicit

Quick-Pick Control

Controls the mouse action when applied within the graphics area.

Manipulation Tools

Provides access to PRIMER specific functions.

Keywords

This provides access to the Keywords that are supported and can be edited by PRIMER.

Graphics Area

Area within which graphics are drawn.

Open Menu tabs

These control which option is displayed in the current menu panel. Model and Part Tree will always be available in addition to selected options.

Current menu panel

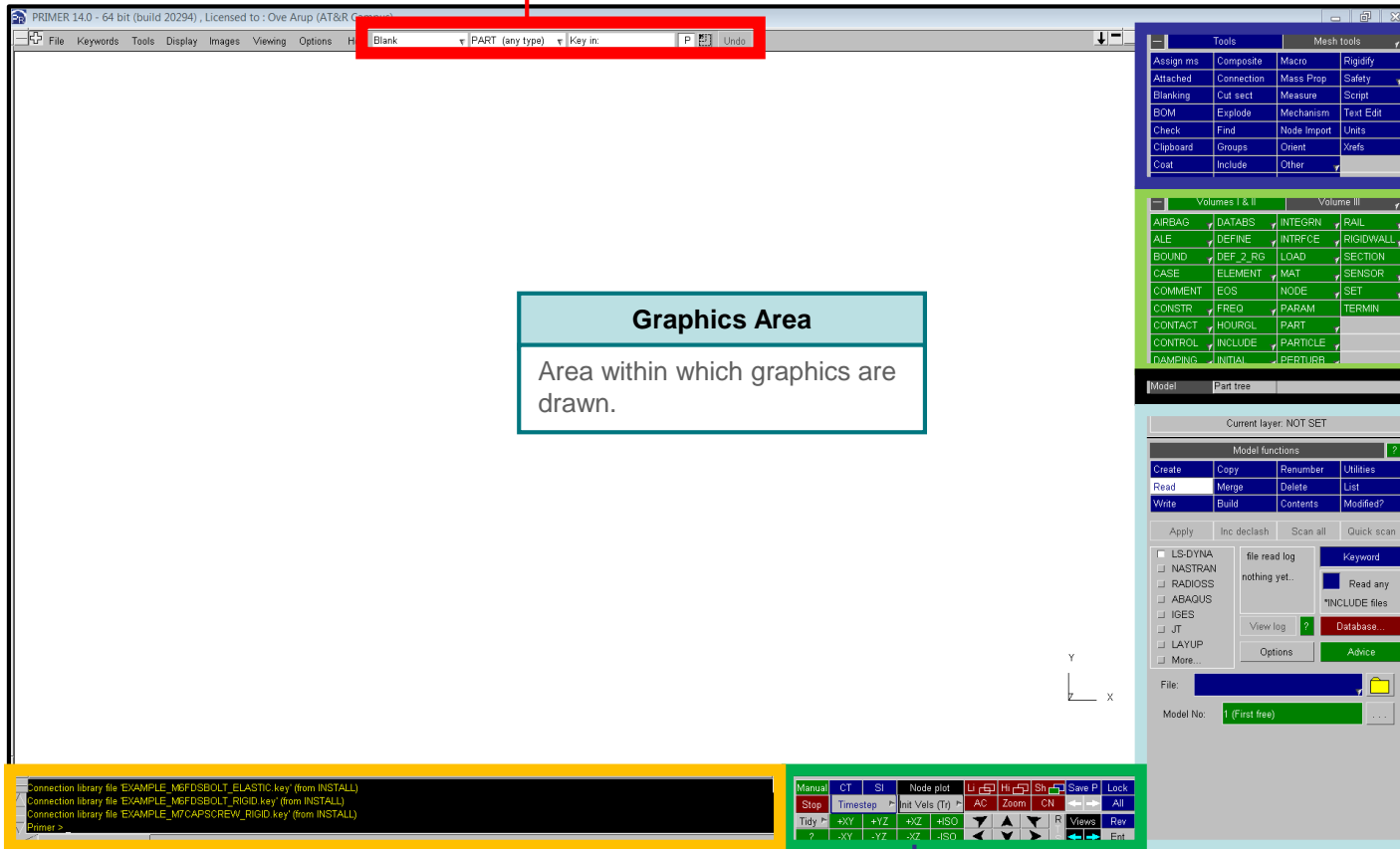
"Current Menu Panel" Displays the menu for the option currently selected by the menu tabs.

Dialogue & List area

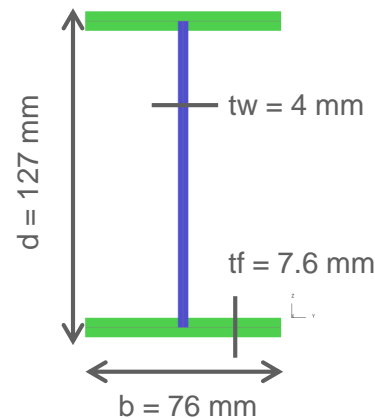
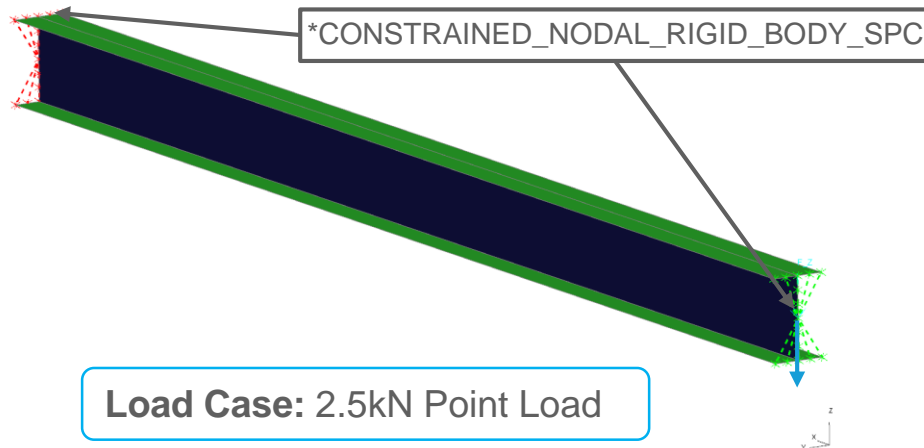
Area for command-line input and output, also acts a listing area for messages.

Viewing & Drawing Commands

Provides all aspects of view control: direction, perspective, scale, etc. Contains the drawing commands and their settings.



Cantilever beam (I section) modelled in shell elements, fixed at one end



$E = 205000 \text{ N/mm}^2$
 $L = 2000 \text{ mm}$
 $\text{Density} = 7.85e-9 \text{ T/mm}^3$
 $\text{Pr} = 0.3$
 $I_y = 46902000 \text{ mm}^4$

Hand calculation: Tip Deflection

$$d = \frac{PL^3}{3EI} \left(1 + \frac{3f_s EI}{GAL^2} \right) \text{ where } f_s = \frac{A}{A_{web}}$$

Hand calculation: Max Bending Moment

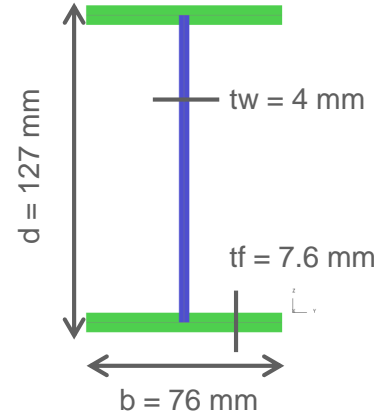
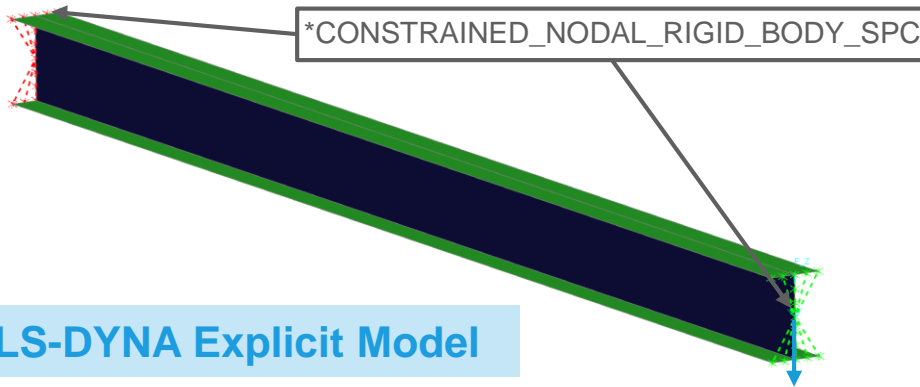
$$M_y = P \cdot L = 2500 \cdot 2000 = 5e6 \text{ Nmm}$$

$$d_z = \frac{PL^3}{3EI_y} \left(1 + \frac{3f_s EI_y}{GAL^2} \right) = \frac{2.5 \cdot 2000^3}{3 \cdot 20500 \cdot 4690200} \left(1 + \frac{3 \cdot (1632.8/477.6) \cdot 20500 \cdot 4690200}{7.88e^4 \cdot 1632.8 \cdot 2000^2} \right) = 13.9 \cdot (1 + 0.019) = 7.07 \text{ mm}$$

$$G = \frac{E}{2(1+\nu)}$$

Model Description: Cantilever Beam Analysis

LS-DYNA Implicit



$E = 205000 \text{ N/mm}^2$
 $L = 2000 \text{ mm}$
 $\text{Density} = 7.85\text{e-}9 \text{ T/mm}^3$
 $\text{Pr} = 0.3$
 $P = 2500 \text{ N}$
 $I_y = 46902000 \text{ mm}^4$

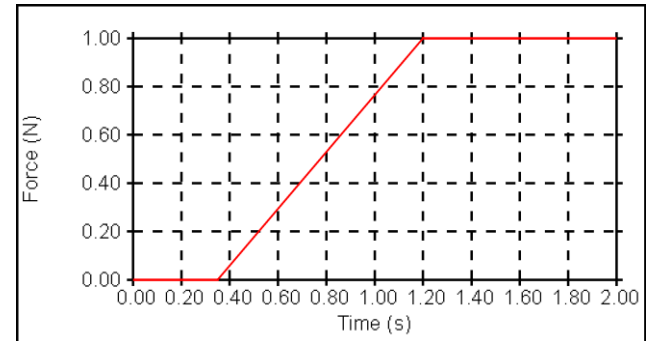
LS-DYNA Explicit Model

```

*MAT_ELASTIC_TITLE
Steel
$:      mid      ro      e      pr      da      db      k
        1      7.85E-9  205000.0  0.3      0.0      0.0      0.0

*SECTION_SHELL
$:      label    elform    shrf      nip      propt      qr      icomp      setyp
        1          16        0.0        5        1.0        0.0        0        0
$:      t1      t2      t3      t4      nloc      marea      idof      edgset
        7.6      7.6      7.6      7.6        0.0        0.0        0.0        0
$
$:      label    elform    shrf      nip      propt      qr      icomp      setyp
        2          16        0.0        5        1.0        0.0        0        0
$:      t1      t2      t3      t4      nloc      marea      idof      edgset
        4.0      4.0      4.0      4.0        0.0        0.0        0.0        0

*PART
Flange
$:      pid      secid      mid      eosid      hgid      grav      adpopt      tmid
        2          1          1          0          0          0          0          0
$
*PART
Web
$:      pid      secid      mid      eosid      hgid      grav      adpopt      tmid
        3          2          1          0          0          0          0          0
    
```



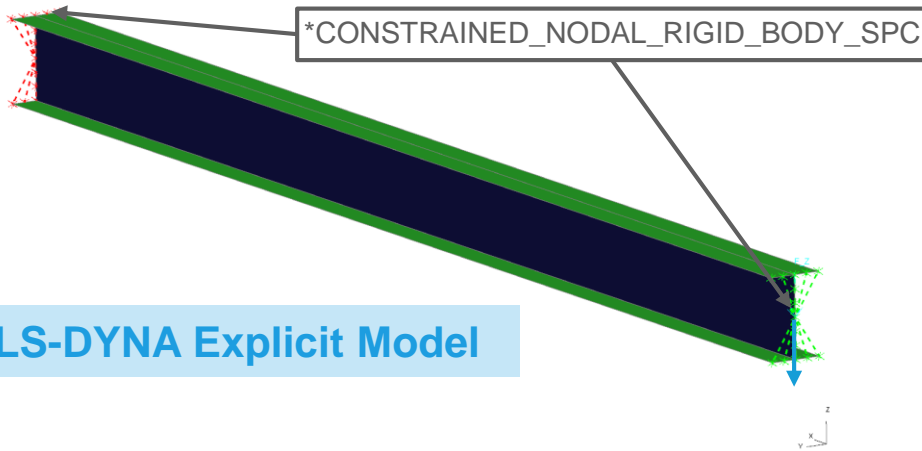
↑ LCID

```

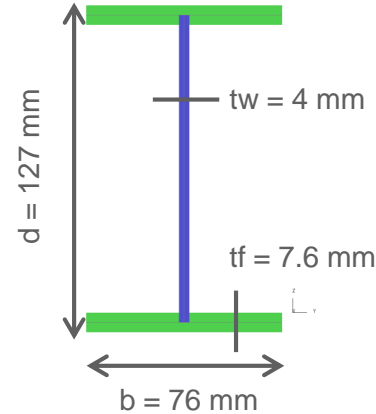
*LOAD_NODE_POINT
$:nid/nsid      dof      lcid      sf
        1277        3        1      -2500.0
    
```

Model Description: Cantilever Beam Analysis

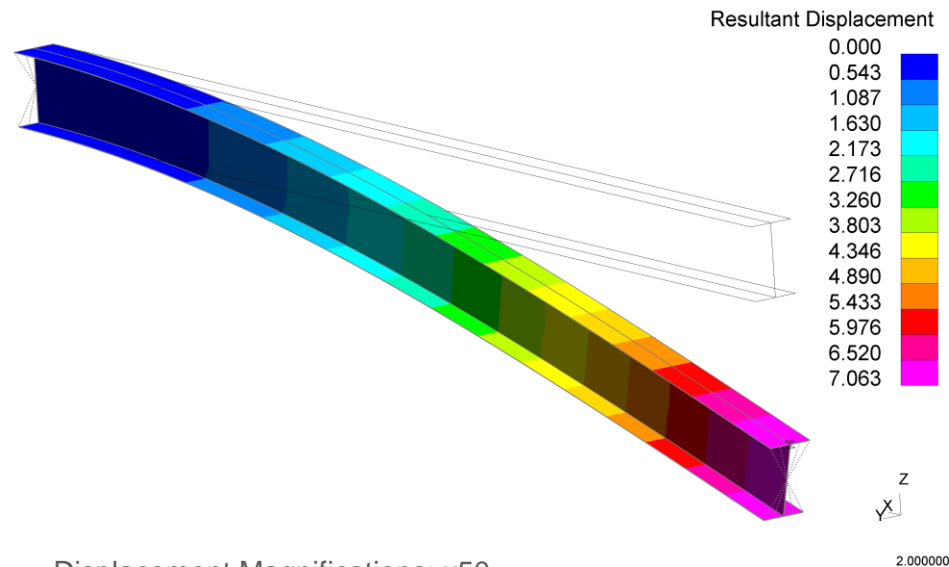
LS-DYNA Implicit



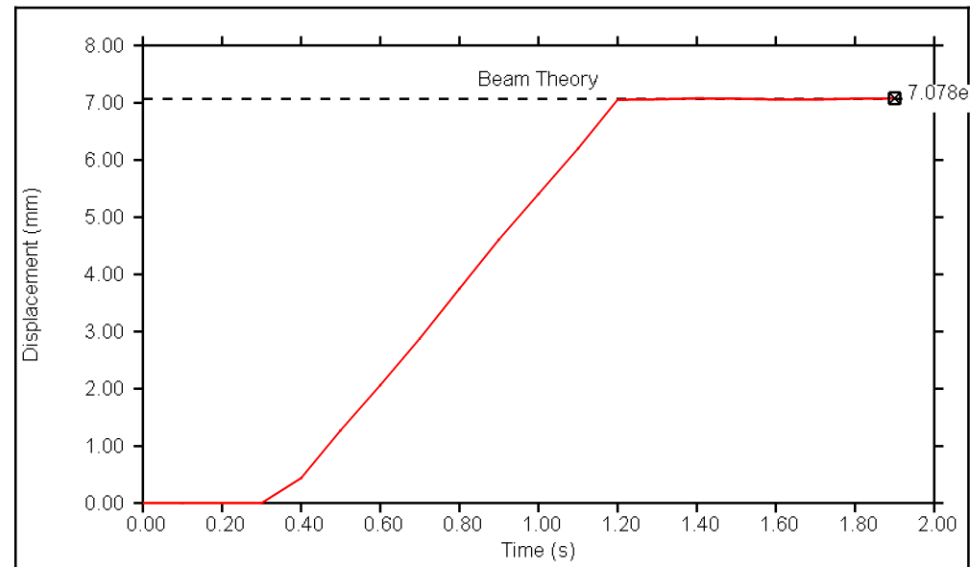
LS-DYNA Explicit Model



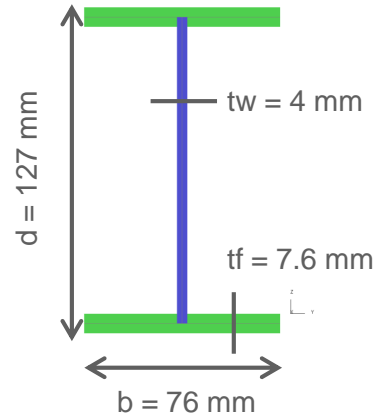
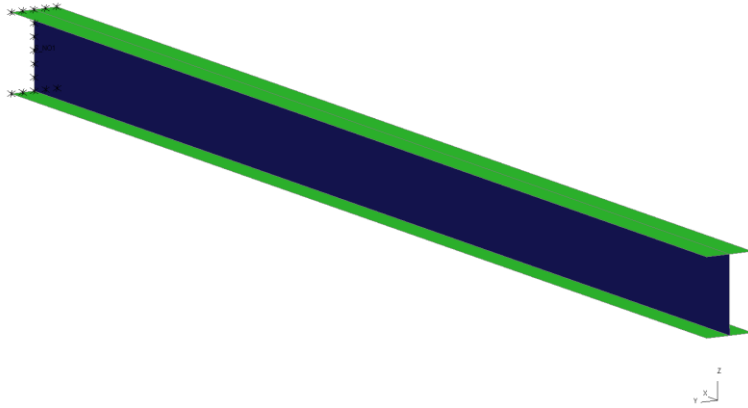
$E = 205000 \text{ N/mm}^2$
 $L = 2000 \text{ mm}$
 $\text{Density} = 7.85\text{e-}9 \text{ T/mm}^3$
 $\text{Pr} = 0.3$
 $P = 2500 \text{ N}$
 $I_y = 46902000 \text{ mm}^4$



Displacement Magnifications: x50



Recommended steps



$E = 205000 \text{ N/mm}^2$
 $L = 2000 \text{ mm}$
 $\text{Density} = 7.85\text{e-}9 \text{ T/mm}^3$
 $Pr = 0.3$
 $P = 2500 \text{ N}$
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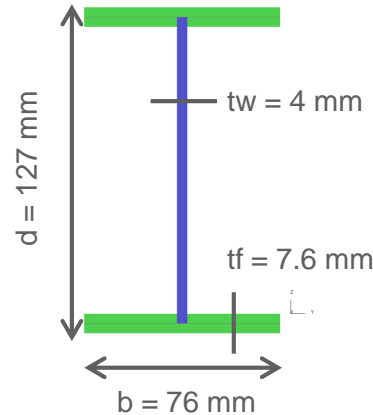
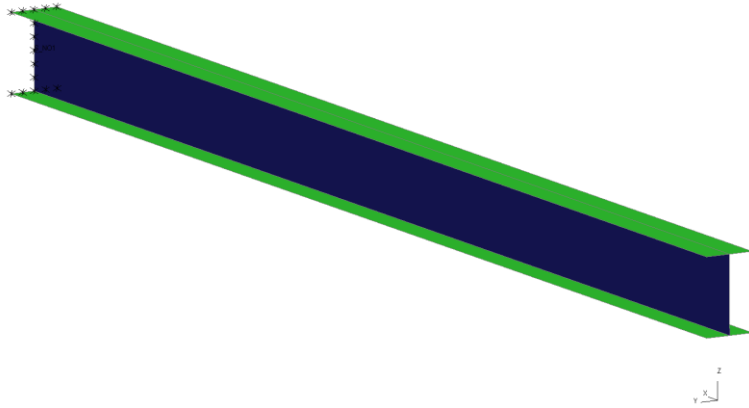
Submit eigenvalue analysis to check and eliminate rigid body modes

- Set up recommended ***CONTROL_**, ***DATABASE_**, and ***SECTION** cards for implicit analysis
- Set-up specific eigenvalue analysis ***CONTROL_** cards
- Submit the analysis
- Check the results and compare against analytical solution

Submit linear static analysis

- Setup implicit load case
- Submit the analysis
- Check the results and compare against beam theory

Recommended steps



$E = 205000 \text{ N/mm}^2$
 $L = 2000 \text{ mm}$
 $\text{Density} = 7.85\text{e-}9 \text{ T/mm}^3$
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Submit linear static analysis

- Setup implicit load case
- Submit the analysis
- Check the results and compare against beam theory

Eigenvalue Analysis

For an implicit static analysis to solve there must be no rigid body modes in the model. If rigid body modes are present these must be eliminated by adding suitable restraints to the model. For this reason the first step in running a static analysis should be an eigenvalue analysis. Capturing the first 20 modes is more than sufficient.

Disable all ***CONTROL** cards in explicit input deck and activate the following:

```
$
*CONTROL_IMPLICIT_GENERAL
$: imflag      dt0      imform      nsbs      igs      cnstn      form      zero_v
   1          0.0        0          0          0          0          0          0
$
*$CONTROL_IMPLICIT_EIGENVALUE
$: neig      center      lflag      lftend      rflag      rhtend      eigmeth      shfscl
   20        0.0          0          0.0          0          0.0          0          0.0
$: isolid      ibeam      ishell      itshell      mstres      evdump      mstrscl
   0            0          0          0          0          0          0.0
*DATABASE_FORMAT
$: iform      ibinary
   0            1
$
*$DATABASE_BINARY_D3PLOT
$: dt          lcdt      nobeam      npltc      psetid
   0.1         0          0          0          0
```

Turn on implicit solver

Set number of eigenvalues to solve for

32 bit **output files**

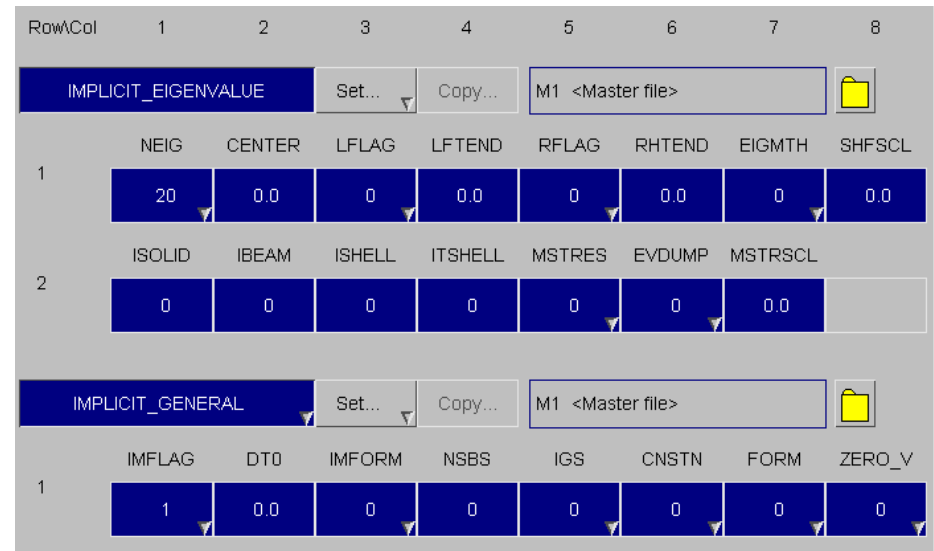
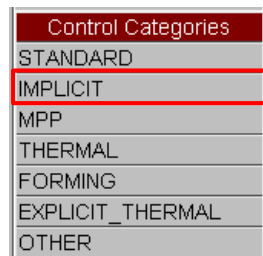
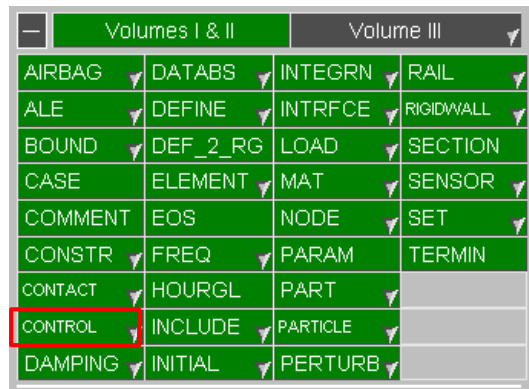
Set a higher **D3PLOT output** than you would for an explicit analysis. Make sure you capture any key points in the analysis.

Eigenvalue Analysis

For an implicit static analysis to solve there must be no rigid body modes in the model. If rigid body modes are present these must be eliminated by adding suitable restraints to the model. For this reason the first step in running a static analysis should be an eigenvalue analysis. Capturing the first 20 modes is more than sufficient.

Disable all ***CONTROL** cards in explicit input deck and activate the following implicit cards:

- Click '**CONTROL**' in the Oasys PRIMER Keyword panel and then '**Modify**'
- Under '**Control Categories**', select '**IMPLICIT**'



Particular element formulations (**ELFORM** on the ***SECTION** cards) are recommended for implicit analyses to give the most accurate results.

Linear Analyses (including eigenvalue)

- Solid Elements – Element formulation type 18 (20, 21, 99)
- **Shell Elements – Element formulation type 18 (20, 21, 99)**
- Beam Elements – Element formulation type 13

MODIFY SECTION M1/SECT1

Buttons: Abort Modify, Restore Original, Text edit, Update SECTION, Copy Existing, Sketch, View Xrefs, Check Defn, Only

Include: M1 <Master file>

Modify SECTION section 1 (model 1)

Label: 1 Title: <No section name given>

Type: SHELL

Row\Col	1	2	3	4	5	6	7	8
1	SECID	ELFORM	SHRF	NIP	PROPT	QR/IRID	ICOMP	SETYP
1	1	18	0.0	5	1.0	0.0	0	0
2	T1	T2	T3	T4	NLOC	MAREA	IDOF	EDGSET
	7.6	7.6	7.6	7.6	0.0	0.0	0.0	0



Volumes I & II Volume III

AIRBAG DATABS INTEGRN RAIL

ALE DEFINE INTRFCE RIGDWALL

BOUND DEF_2_RG LOAD SECTION

CASE ELEMENT MAT SENSOR

COMMENT EOS NODE SET

CONSTR FREQ PARAM TERMIN

CONTACT HOURGL PART

CONTROL INCLUDE PARTICLE

DAMPING INITIAL PERTURB

Model Part tree L_NODE CURVE

Section

M1.Main file

Section

Create Delete Check Help

Copy Sketch Renumbr

Modify Keyword List

Apply

MODIFY section M1/SECT1

SELECT SECTION

All None Opt

Filter Vis key_In Sk

(M/L) SECTION(s) (all mo

M1/SECT1 (SHELL)

M1/SECT2 (SHELL)

Rather than modifying an existing explicit model the **ISOLID, IBEAM, ISHELL, ITSHELL** options in the ***CONTROL_IMPLICIT_EIGENVALUE** card can be used to automatically reset the element type.

Note: As long as **NEIG=0** on the ***CONTROL_IMPLICIT_EIGENVALUE** card LS-DYNA will not run an eigenvalue analysis.

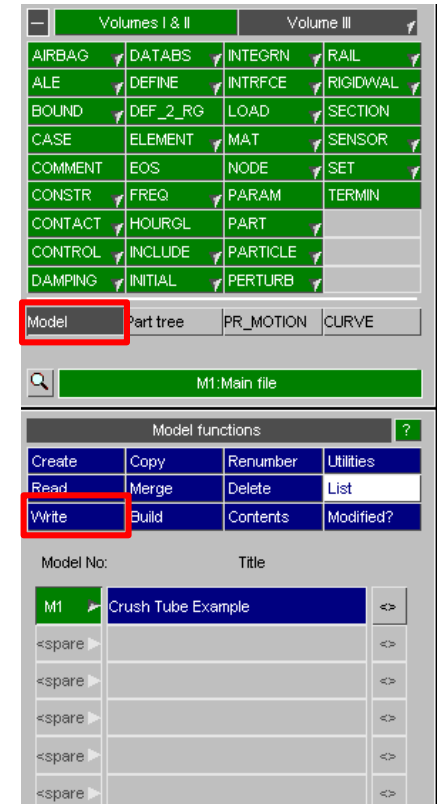
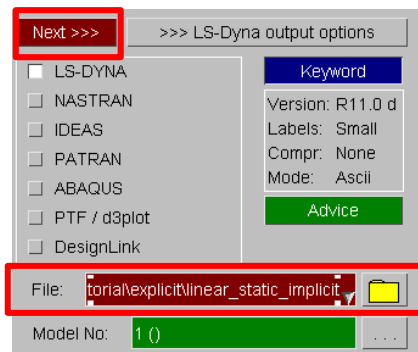
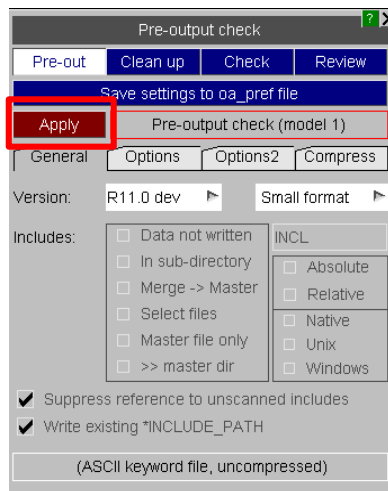
Example

***CONTROL_IMPLICIT_EIGENVALUE** settings for switching all solid elements to formulation 23 and all shell elements to formulation 16:

The screenshot shows the 'CONTROL' dialog box in LS-DYNA. The 'IMPLICIT_EIGENVALUE' card is selected. The 'NEIG' field is set to 0. The 'ISOLID' field is set to 23, and the 'ISHELL' field is set to 16. Both 23 and 16 are highlighted with red boxes. The 'M1 <Master file>' field is also visible.

	NEIG	CENTER	LFLAG	LFTEND	RFLAG	RHTEND	EIGMTH	SHFSCL
1	0	0.0	0	0.0	0	0.0	0	0.0
2								
	ISOLID	IBEAM	ISHELL	ITSHELL	MSTRES	EVDUMP	MSTRSCL	
	23	0	16	0	0	0	0.0	

- To save the model click '**Model**' and then '**Write**'.
- Define a file path and name (**linear_static_eigv.key**) and then click '**Next**' and finally click '**Apply**', to write the file.
- The written keyword file can now be submitted to LS-DYNA.

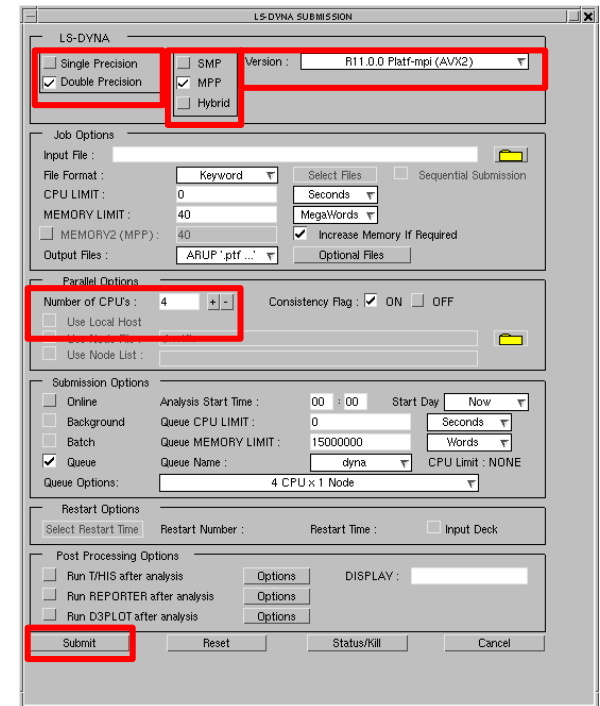
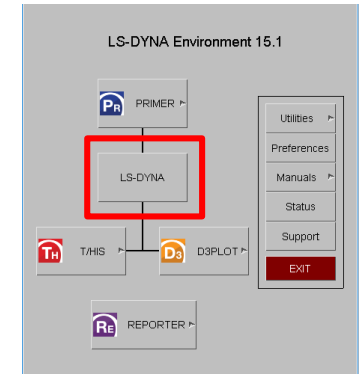


Solver: Submit LS-DYNA job

- In Oasys SHELL, select the following settings:
 - Double Precision:** Double precision (DP) use 64 bit storage and is useful when extremely small deflections or long event times are expected. Recommended for implicit analyses.
 - MMP:** Massively Parallel Processing (MPP) allows users to run the LS-DYNA solver over a cluster of machines or use multiple processors on a single machine. Recommended for implicit analyses.
 - Version:** Select the desired SMP LS-DYNA version installed on the client machine.
 - Number of CPUs:** 4 cores is sufficient here.
 - Input File:** Navigate to the LS-DYNA input deck
 - Click 'Submit'.

For best implicit performance, it is important to provide enough memory to allow the stiffness matrix factorization to run in-core

LS-DYNA Implicit



Run the analysis in LS-DYNA and check for rigid body (zero frequency) modes in the “eigout” file, see below.

Check for frequencies at or near zero

problem time = 1.00000E+00

(all frequencies de-shifted)

MODE	EIGENVALUE	----- frequency -----		
		RADIANS	CYCLES	PERIOD
1	6.915707E+03	8.316073E+01	1.323544E+01	7.555472E-02
2	5.298313E+04	2.301807E+02	3.663439E+01	2.729676E-02
3	5.618522E+04	2.370342E+02	3.772517E+01	2.650750E-02
4	2.678063E+05	5.175000E+02	8.236269E+01	1.214142E-02
5	8.222794E+05	9.067963E+02	1.443211E+02	6.928993E-03
6	1.842911E+06	1.357539E+03	2.160590E+02	4.628366E-03
7	2.029133E+06	1.424476E+03	2.267125E+02	4.410873E-03
8	3.998877E+06	1.999719E+03	3.182652E+02	3.142034E-03
9	6.946053E+06	2.635537E+03	4.194587E+02	2.384025E-03
10	7.207740E+06	2.684723E+03	4.272870E+02	2.340347E-03



λ

$\omega = \sqrt{\lambda}$

$f = \omega / 2\pi$

$T = 1 / f$

The modes are output to the “d3eigv” file which can be animated in Oasys D3PLOT.

Post-processing: Oasys D3PLOT

LS-DYNA Implicit

Top menus

Allows access to basic options, keywords and tools, in a drop-down menu format

Quick-Pick Control

Controls the mouse action when applied within the graphics area

Tools

Provides access to D3PLOT specific functions

T/HIS link

Menu tabs

These control which option is displayed in the current menu panel. Model and Part Tree will always be available in addition to selected options

Current menu panel

"Current Menu Panel" Displays the menu for the option currently selected by the menu tabs

Animation Controls

Controls states and what is displayed during animation

Window ID

Graphics Area

Area within which graphics are drawn

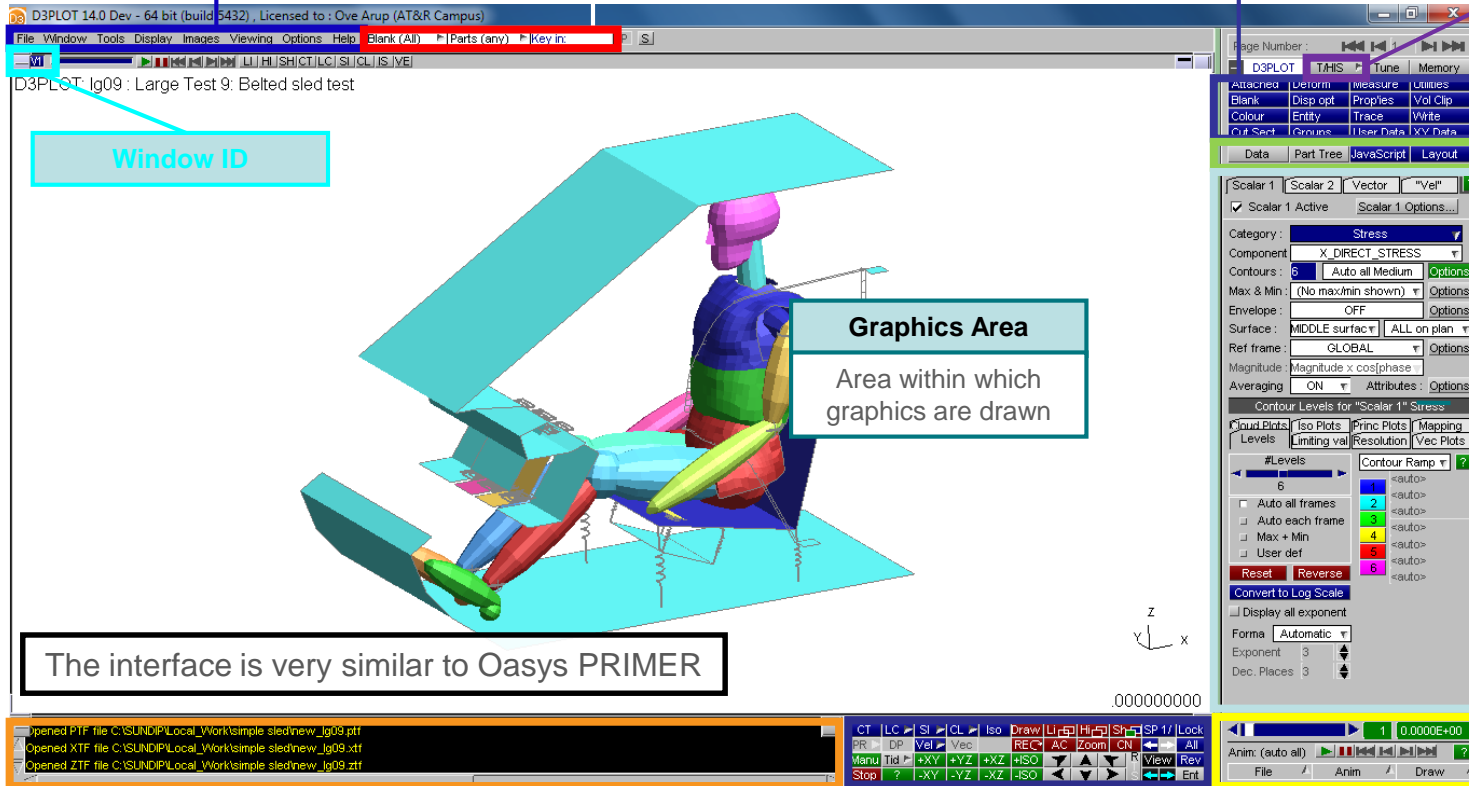
The interface is very similar to Oasys PRIMER

Dialogue & List area

Area for command-line input and output, also acts a listing area for messages

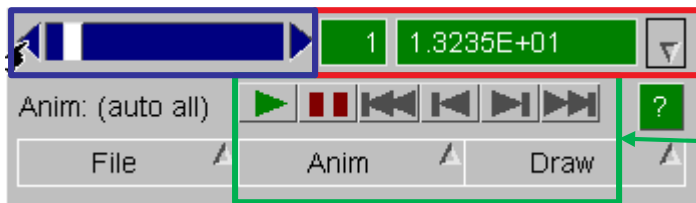
Viewing & Drawing Commands

Provides all aspects of view control: direction, perspective, scale, etc. Contains the drawing commands and their settings



- Open the d3eigv file from the run directory in Oasys D3PLOT.
- Note: Eigenvalue analysis is a special and the usual d3plot/ptf file will not display any results
- Animate the results and check the mode shapes of the beam.

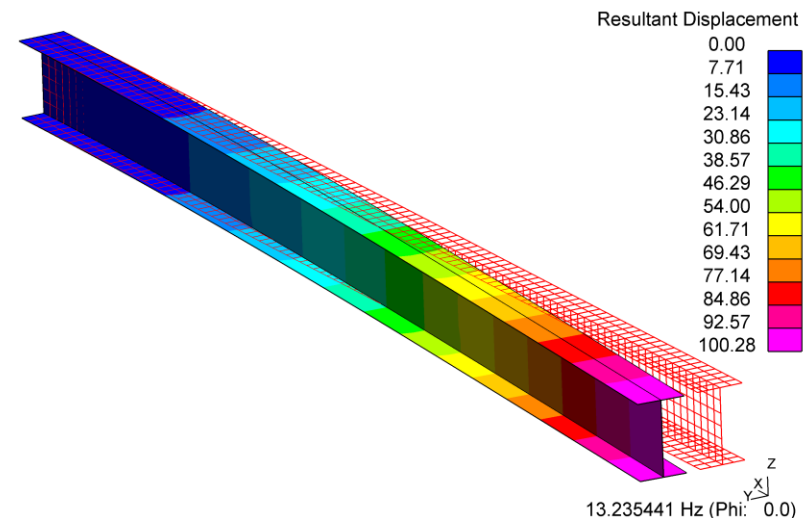
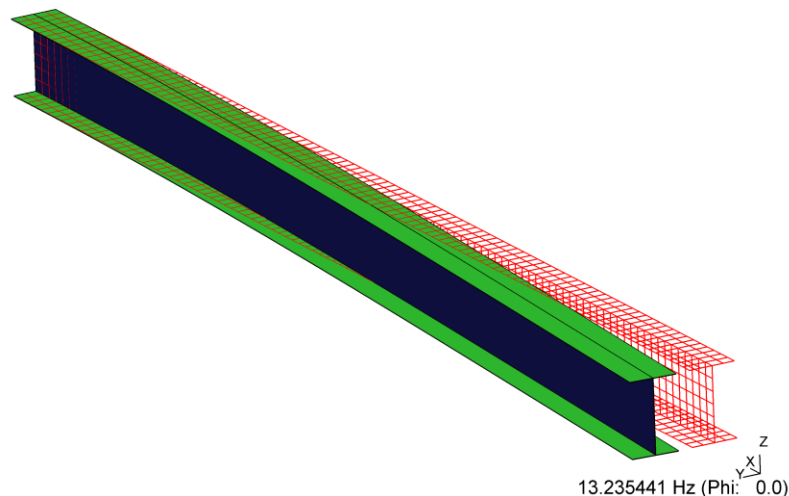
Position the slide on the state number



Natural frequency

Animate results

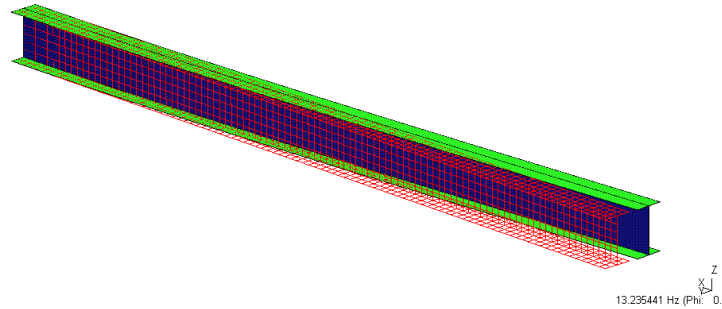
Contour plots



No Rigid Body Modes

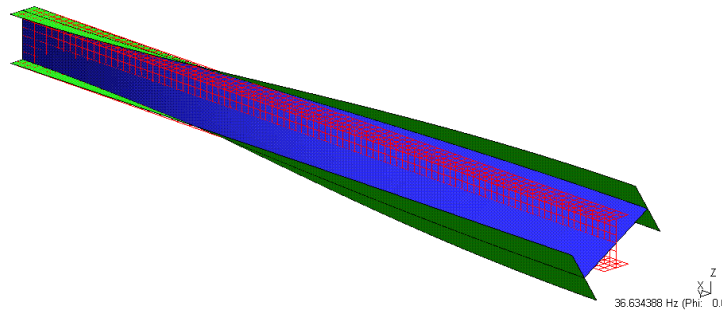
This is because the beam has been suitably restrained.
(Try removing the SPC: rigid modes will appear)

D3PLOT: LS-DYNA eigenvalues at time 1.00000E+00



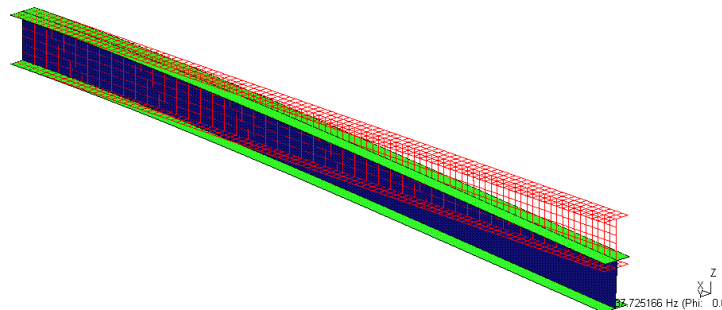
Mode 1 – 13Hz

D3PLOT: LS-DYNA eigenvalues at time 1.00000E+00

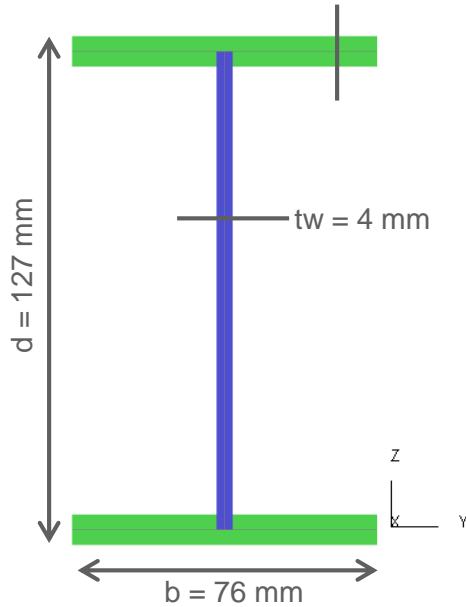


Mode 2 – 37Hz

D3PLOT: LS-DYNA eigenvalues at time 1.00000E+00



Mode 3 – 38Hz



Hand calculation: 1st Natural Frequency

$$f_1 = \frac{1.875^2 \sqrt{\frac{EI}{\rho A l^4}}}{2\pi}$$

$$E = 205000 \text{ N/mm}^2$$

$$L = 2000 \text{ mm}$$

$$\text{Density} = 7.85e-9 \text{ T/mm}^3$$

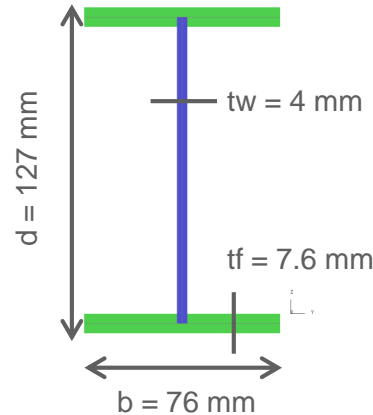
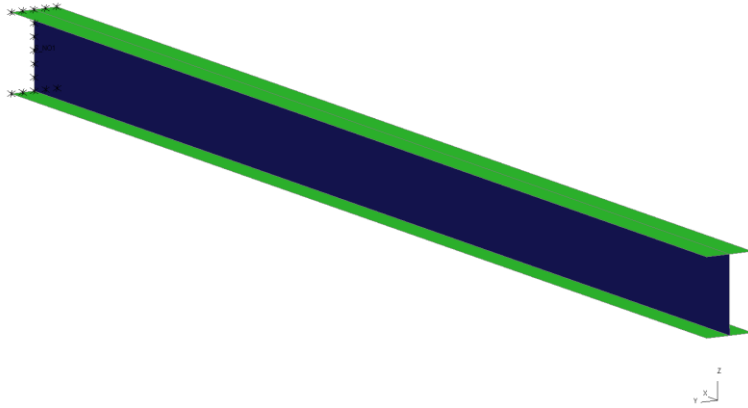
$$I_y = \sum (I_{yi} + A d_z^2) = \frac{1}{12} 4 \cdot 119.4^3 + 2 \cdot \left(\frac{1}{12} 76 \cdot 7.6^3 + (7.6 \cdot 76) \left(\frac{119.4}{2} \right)^2 \right) = 4690200 \text{ mm}^4$$

$$f_1 = \frac{1.875^2 \sqrt{\frac{EI_y}{\rho A l^4}}}{2\pi} = \frac{1.875^2 \sqrt{\frac{205000 \cdot 4690200}{7.85e-9 \cdot 1632.8 \cdot 2000^4}}}{2\pi} = 38.3 \text{ Hz} \checkmark$$

$$I_z = \sum (I_{zi} + A d_y^2) = \frac{1}{12} 119.4 \cdot 4^3 + 2 \cdot \left(\frac{1}{12} 7.6 \cdot 76^3 \right) = 556673.07 \text{ mm}^4$$

$$f_1 = \frac{1.875^2 \sqrt{\frac{EI_z}{\rho A l^4}}}{2\pi} = \frac{1.875^2 \sqrt{\frac{205000 \cdot 556673.07}{7.85e-9 \cdot 1632.8 \cdot 2000^4}}}{2\pi} = 13.2 \text{ Hz} \checkmark$$

Recommended steps



$E = 205000 \text{ N/mm}^2$
 $L = 2000 \text{ mm}$
 $\text{Density} = 7.85\text{e-}9 \text{ T/mm}^3$
 $Pr = 0.3$
 $P = 2500 \text{ N}$
 $I_y = 46902000 \text{ mm}^4$

Submit eigenvalue analysis to check and eliminate rigid body modes

- Set up recommended *CONTROL_, *DATABASE_, and *SECTION cards for implicit analysis
- Set-up specific eigenvalue analysis *CONTROL_ cards
- Submit the analysis
- Check the results and compare against analytical solution

Submit linear static analysis

- Setup implicit load case and boundary conditions
- Submit the analysis
- Check the results and compare against beam theory

When running a **static analysis** either linear or non-linear it is often useful to request output to the “binout” file so that the boundary forces can be checked against the applied load. If there is an imbalance then the results are likely to be in error.

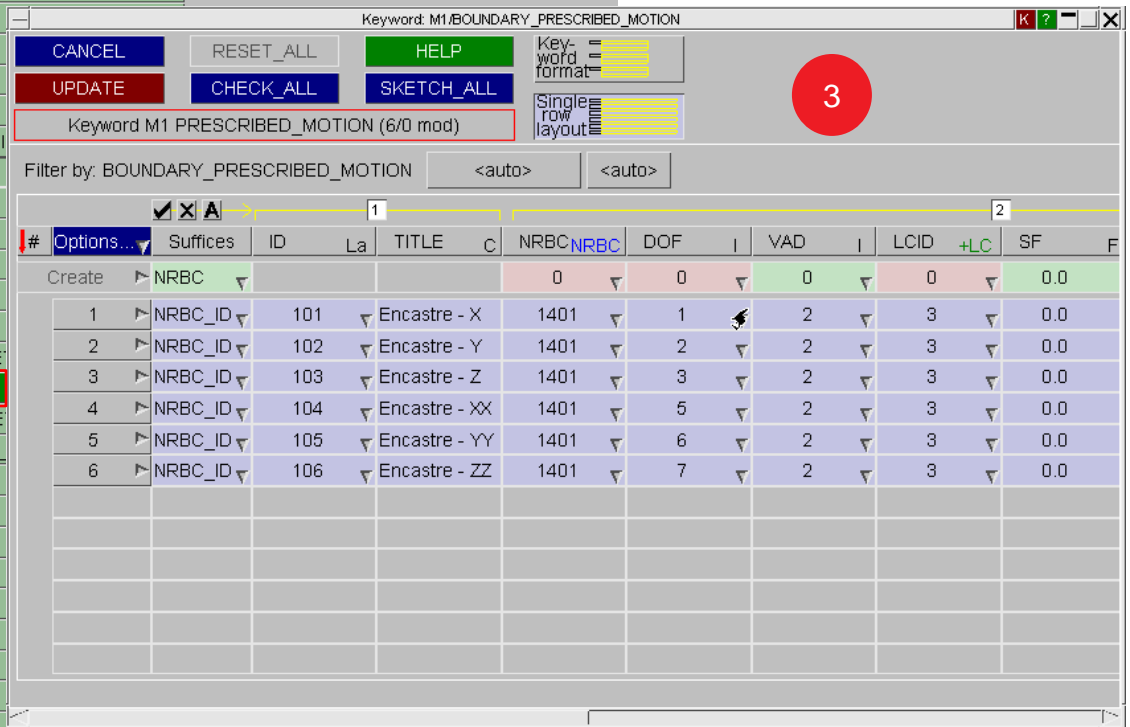
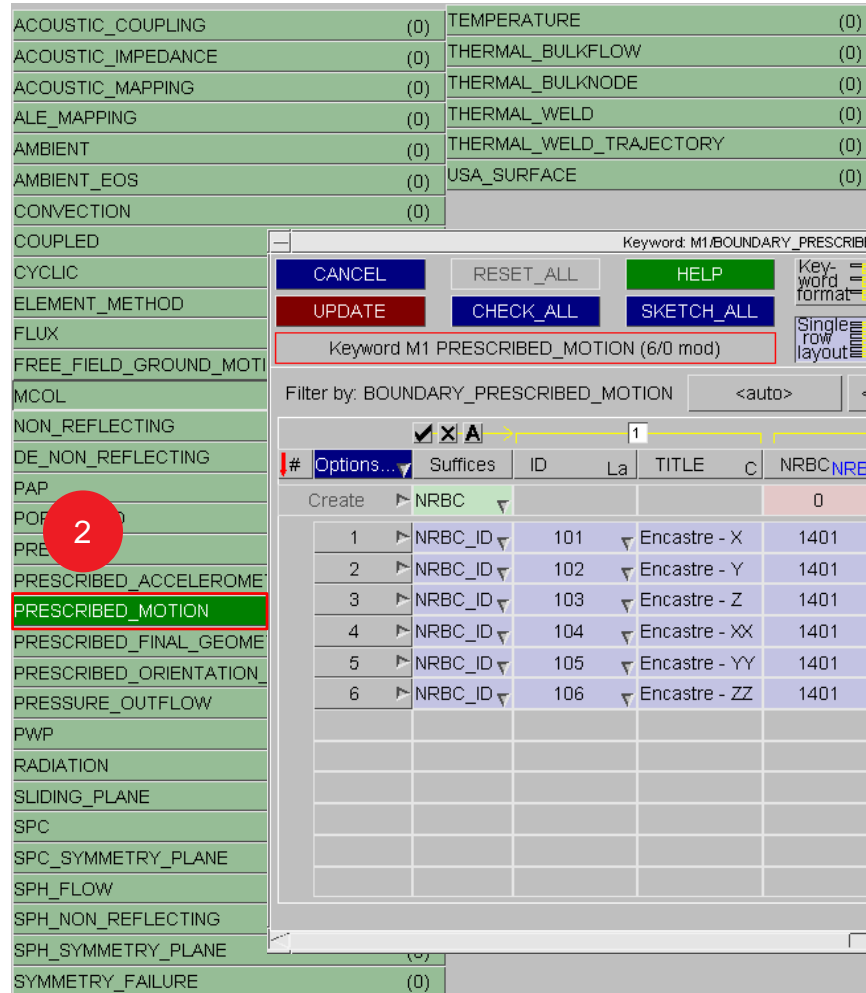
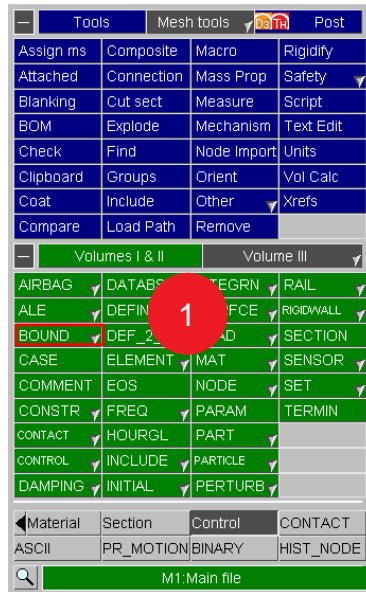
Restrained nodes

- Output to “spcforc” file or
- Output sets of nodes to “nodfor” file

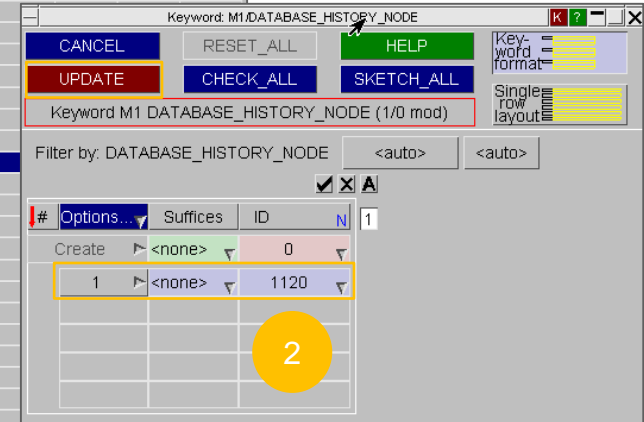
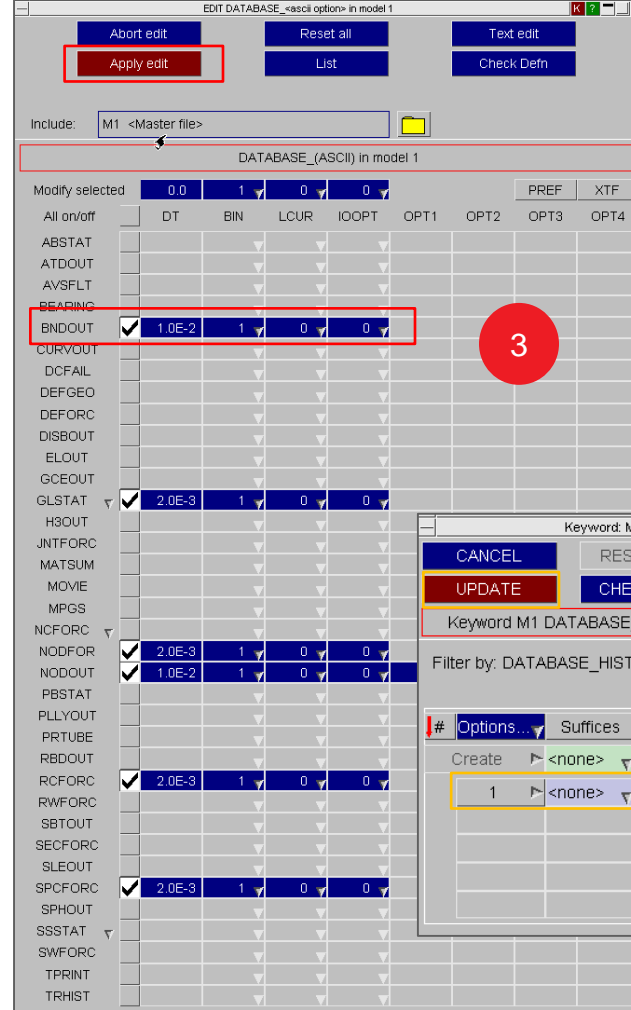
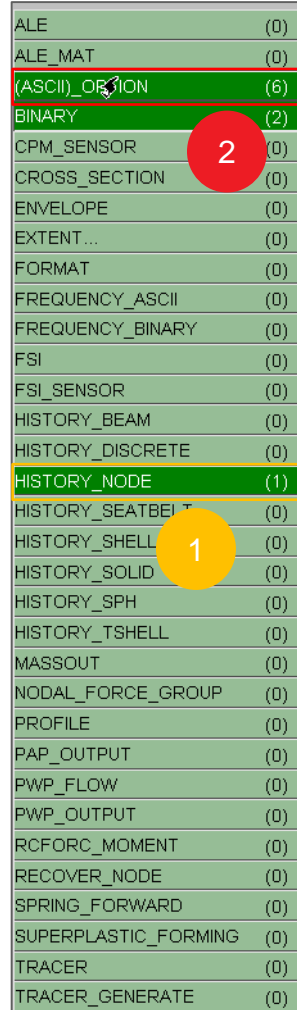
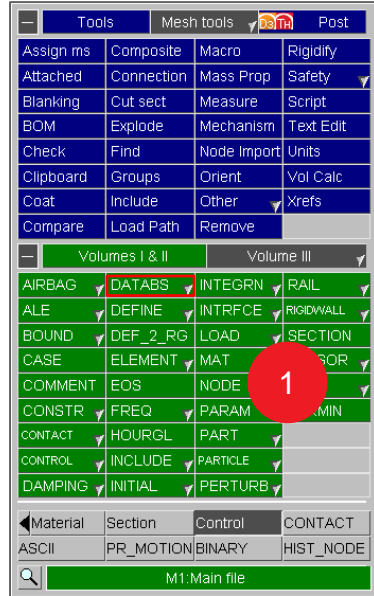
Rigid Parts and Nodal Rigid Bodies

- If these are restrained then at present it is not possible to get the restraining forces output.
- To get these forces the restraints must be replaced by:
 - ***BOUNDARY_PRESCRIBED_MOTION_NRBC** and
 - Setting the “scale factor” to zero
 - Turn on output to the “bndout” file

*BOUNDARY_PRESCRIBED_MOTION_NRBC



*DATABASE (ASCII)_OPTION/*DATABASE_HISTORY_NODE



Linear Static Analysis: Control Cards

LS-DYNA Implicit

```

*CONTROL_ACCURACY
$      OSU      INN      PIDOSU      IACC
      1        2              1

*CONTROL_MPP_IO_NODUMP
*CONTROL_MPP_IO_NOD3DUMP
*CONTROL_MPP_IO_NOFULL
*CONTROL_MPP_IO_LSTC_REDUCE

*CONTROL_ENERGY
$      HGEN      RWEN      SLNTEN      RYLEN
      2          2          2          2

*CONTROL_HOURLGLASS
$      IHQ        QH
      6          0.10

*CONTROL_OUTPUT
$:      npopt      neecho      nrefup      iaccop      opifs      ipnint      ikedit      iflush
      1          3          0          0          0.0          0          0          0
$:      iprtf      ierode      tet10      msgmax      ipcurv      gmdt      ipldblt      eocs
      0          0          0          0          0          0.0          0          0
$:      tolev      newleg      frfreq      minfo      solsig      msgflg      cdetol
      0          0          0          1          0          0          0.0

$=====
*CONTROL_IMPLICIT_GENERAL
$:      imflag      dt0      imform      nsbs      igs      cnstn      form      zero_v
      1          1.0          0          0          0          0          0          0

$
*CONTROL_IMPLICIT_SOLUTION
$:      nsolvr      ilimit      maxref      ddtol      ectol      rctol      lstol      abstol
      1          0          0          0.0          0.0          0.0          0.0          0.0
$:      dnorm      diverg      istif      nlprint      nlnorm      d3itctl      cpchk
      0          0          99999          0          0          0          0

*CONTROL_IMPLICIT_SOLVER
$:      lsolvr      lprint      negev      order      drcm      drcprm      autospc      autotol
      5          2          0          0          0          0.0          2          0.0

$
*CONTROL_TERMINATION
$:      endtim      endcyc      dtmin      endeng      endmas      nosol
      3.0          0          0.0          0.0          0.0          0
    
```

Turn on **objective stress update**,
invariant node numbering and
implicit accuracy.

Remove excessive dump files

Turn on **implicit solver**

Linear solver on: Don't update
stiffness matrix

Use solver 5 – Only solver in
MPP, Get statistics

Termination Time (Load step)
set to 3.0

Open the .otf file and look for: Error, Warning, total, termination, smallest, added
It's useful to look at this file while the model is running. (The *messag* files can be useful too).

The .otf file contains:

- Generic information from reading the model in
- First allocation of memory
- Model initialisation (total masses, smallest timestep, contact surface timesteps (should be less than model timestep), added mass, initial energy, ...)
- Information at each timesteps output files are produced for
- Summary of time taken for each part of the analysis (check it's reasonable).

100 smallest timesteps

```
-----
element      timestep
shell 27156   0.35129E-06
shell 12452   0.35129E-06
shell 30534   0.41070E-06
shell 3       0.41070E-06
slave surface of interface # 1 type= 13
  surface timestep= 0.761E-06 current minimum= 0.761E-06
slave surface of interface # 2 type= 13
  surface timestep= 0.761E-06 current minimum= 0.761E-06
slave surface of interface # 3 type= 7
  surface timestep= 0.498E-06 current minimum= 0.498E-06
master surface of interface # 3 type= 7
  surface timestep= 0.761E-06 current minimum= 0.498E-06

The LS-DYNA time step size should not exceed 0.498E-06
```

```
problem cycle = 1200
time          = 1.2949E-03
added mass    = 4.3696E-04
percentage increase = 2.7420
```

```
Deformable Spotwelds *****
total added spotweld mass = 1.3139E-05
percentage mass increase = 8.2451E-04
```

Timing information

	CPU(seconds)	%CPU	Clock(seconds)	%Clock
Initialization	4.7000E-01	0.01	5.0935E-01	0.01
Element processing ...	4.0769E+03	47.79	2.0127E+03	47.58
Binary databases	3.3946E+00	0.04	1.7572E+00	0.04
ASCII database	2.0024E+00	0.02	1.1212E+00	0.03
Contact algorithm ...	4.3704E+03	51.23	2.1756E+03	51.43
Interface ID 1	3.4023E+03	39.88	1.6947E+03	40.06
Interface ID 2	9.4749E+02	11.11	4.7645E+02	11.26
Contact entities	0.0000E+00	0.00	0.0000E+00	0.00
Rigid bodies	7.7279E+01	0.91	3.8627E+01	0.91
Implicit Nonlinear ...	0.0000E+00	0.00	0.0000E+00	0.00
Implicit Lin. Alg. ...	0.0000E+00	0.00	0.0000E+00	0.00

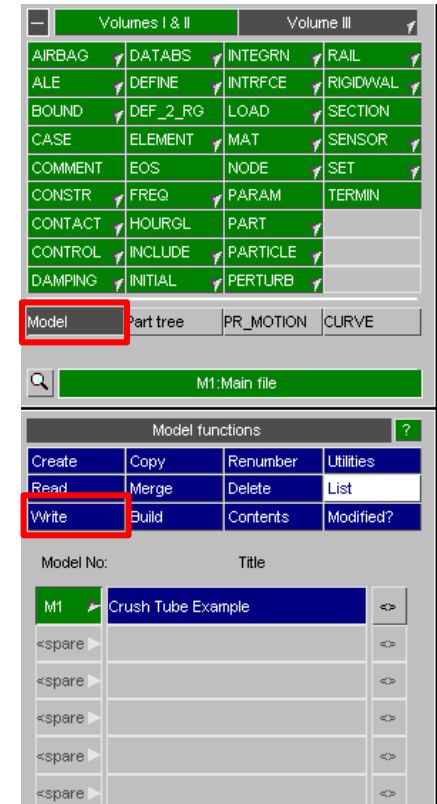
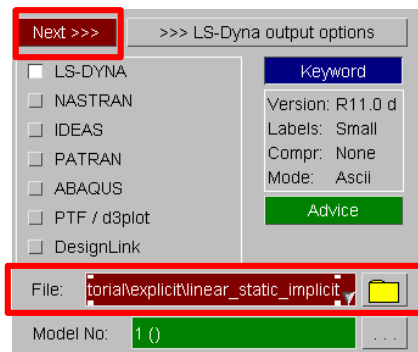
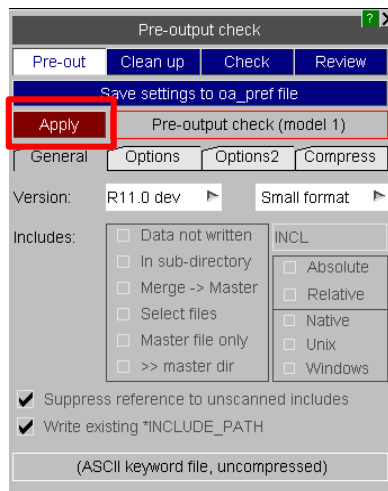
```
Totals      8.5305E+03  100.00  4.2303E+03  100.00
```

```
Problem time      = 5.0000E-01
Problem cycle     = 1171653
Total CPU time    = 8531 seconds ( 2 hours 22 minutes 11 seconds)
CPU time per zone cycle = 810 nanoseconds
Clock time per zone cycle= 402 nanoseconds
```

```
Number of CPU's  2
NLQ used/max     272/ 272
Start time       09/13/2010 16:53:41
End time         09/13/2010 18:04:50
Elapsed time     4269 seconds( 1 hours 11 min. 9 sec.) for 1171653 cycles
```

Normal termination

- To save the model click '**Model**' and then '**Write**'.
- Define a file path and name (**linear_static_implicit.key**) and then click '**Next**' and finally click '**Apply**', to write the file.
- The written keyword file can now be submitted to LS-DYNA.

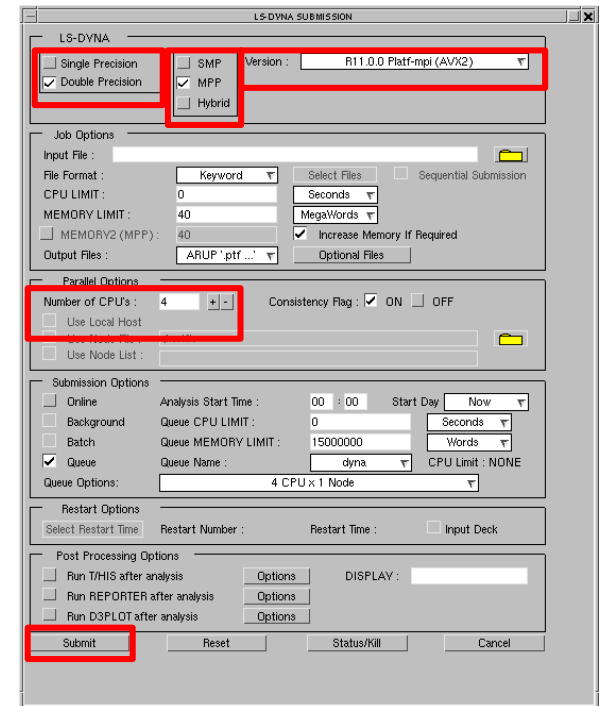
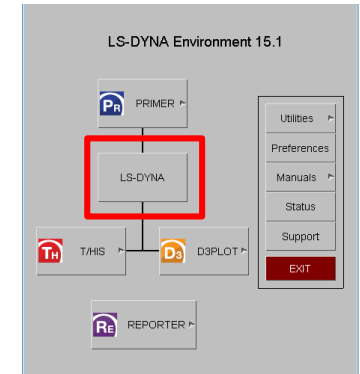


Solver: Submit LS-DYNA job

- In Oasys SHELL, select the following settings:
 - Double Precision:** Double precision (DP) use 64 bit storage and is useful when extremely small deflections or long event times are expected. Recommended for implicit analyses.
 - MMP:** Massively Parallel Processing (MPP) allows users to run the LS-DYNA solver over a cluster of machines or use multiple processors on a single machine. Recommended for implicit analyses.
 - Version:** Select the desired SMP LS-DYNA version installed on the client machine.
 - Number of CPUs:** 4 cores is sufficient here.
 - Input File:** Navigate to the LS-DYNA input deck
 - Click 'Submit'.

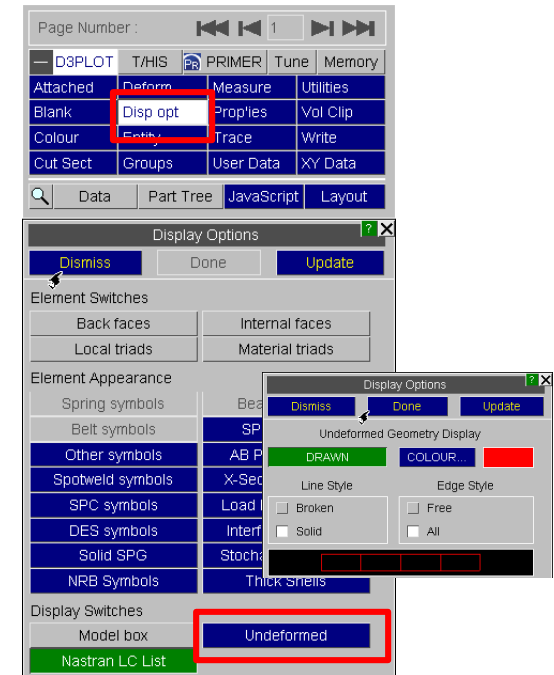
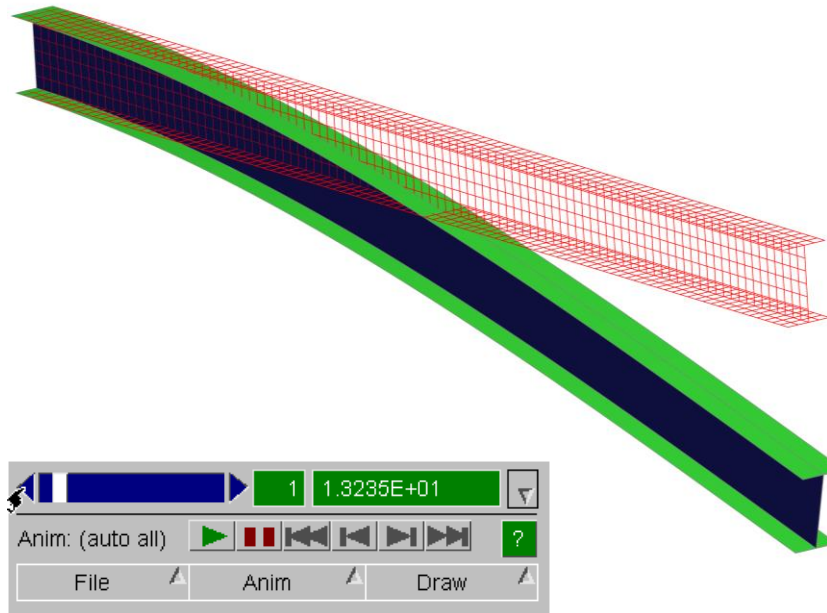
For best implicit performance, it is important to provide enough memory to allow the stiffness matrix factorization to run in-core

LS-DYNA Implicit



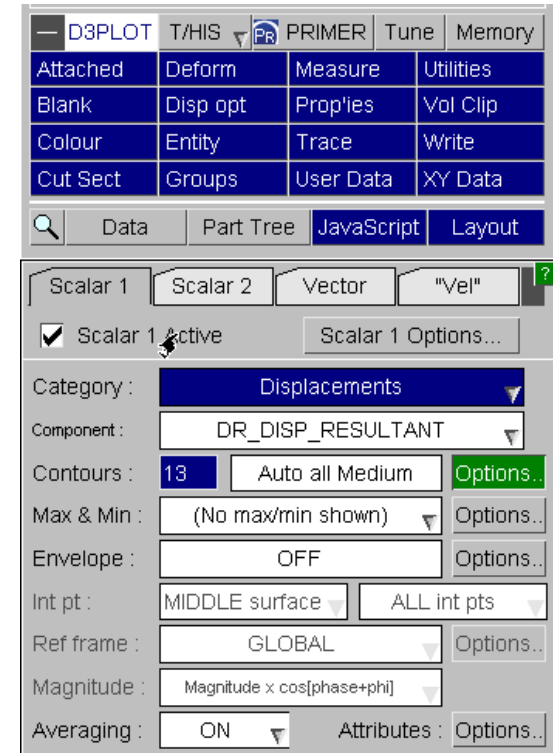
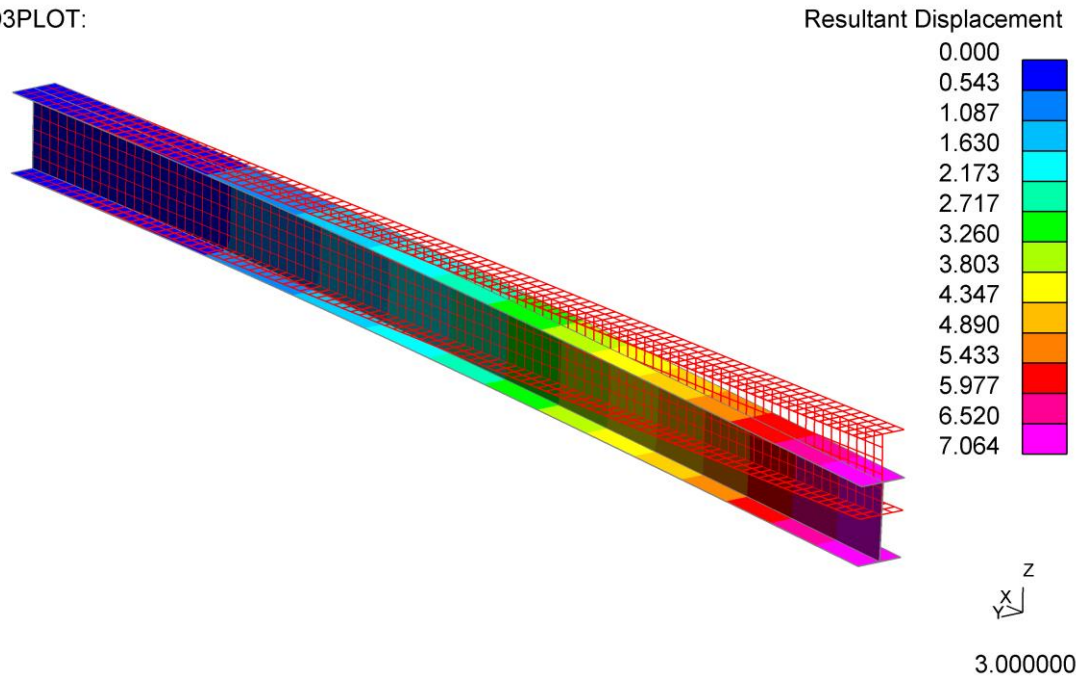
- Open the <name>.ptf file from the run directory in Oasys D3PLOT.
- Left Shift + left mouse button to rotate the camera.
- Clicking on items blanks them. Middle mouse button to unblank last item. 'U' to unblank everything.
- Animate the results and check the deformation.
- Magnify the displacements and display the undeformed geometry.,

DT:



- Select **Category**: Displacements
- **Component**: DR_DISP_RESULTANT via the 'DATA – Scalar 1' panel
- Select **CT** (Continuous Tone) or **SI** (Shaded Image) for contour plotting

D3PLOT:



By default this will show the **Plastic Strain** component, but other output categories can be selected from the data panel.

Top menus

Allows access to basic options and tools, in a drop-down menu format

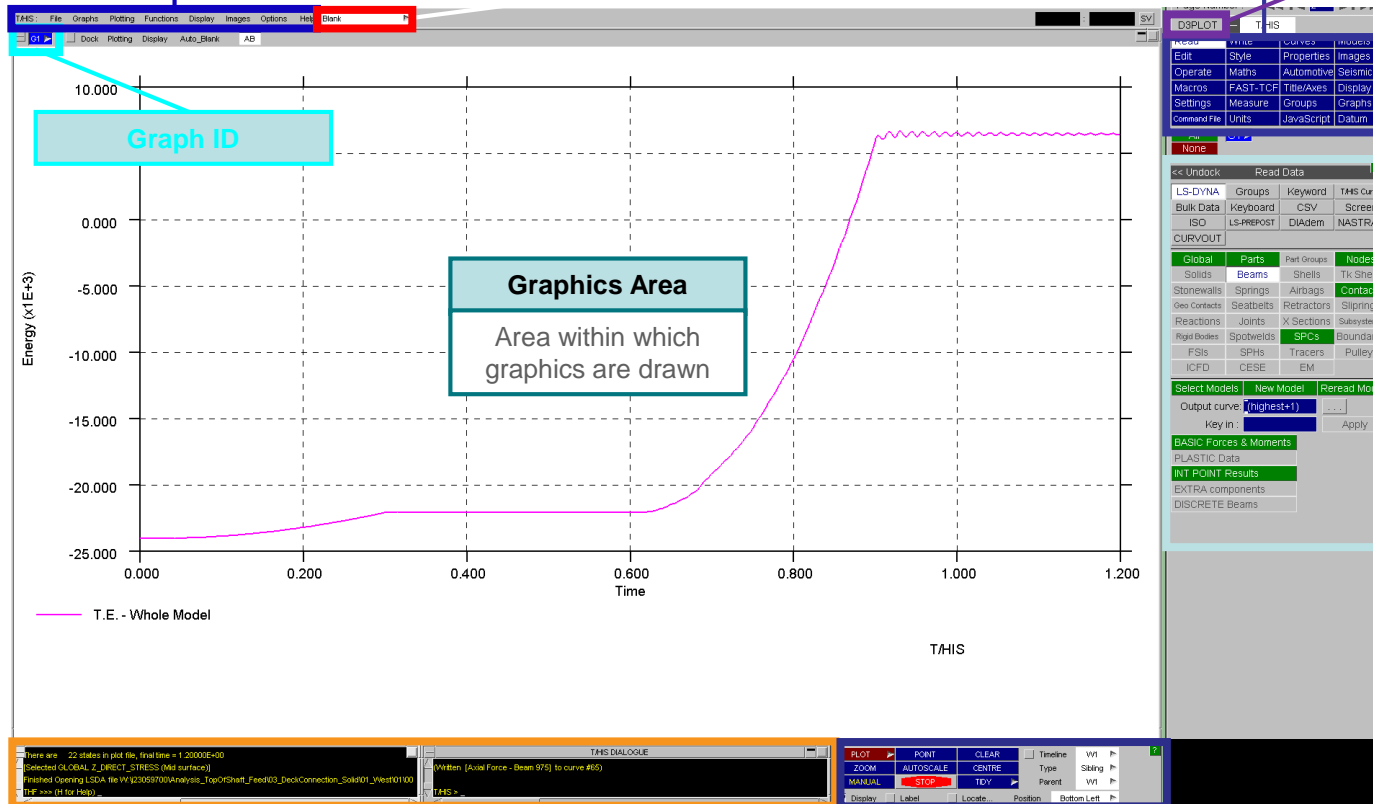
Quick-Pick Control

Controls the mouse action when applied within the graphics area

Tools

Provides access to T/HIS specific functions

D3PLOT Link



Graph ID

Graphics Area

Area within which graphics are drawn

Current menu panel

The entity types for which time-history data is available are shown by the green buttons.

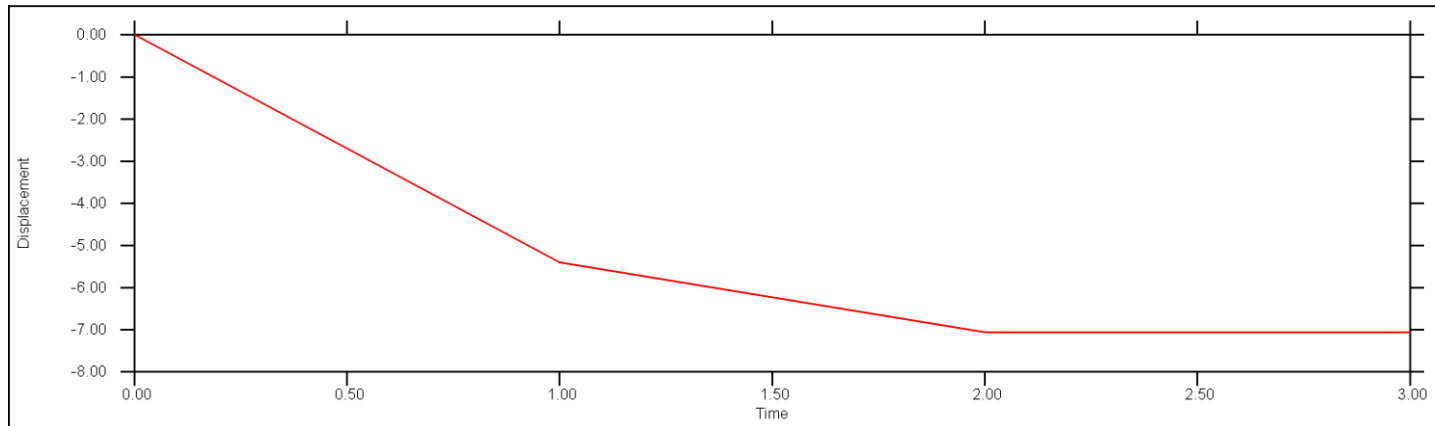
Dialogue & List area

Area for command-line input and output, also acts a listing area for messages

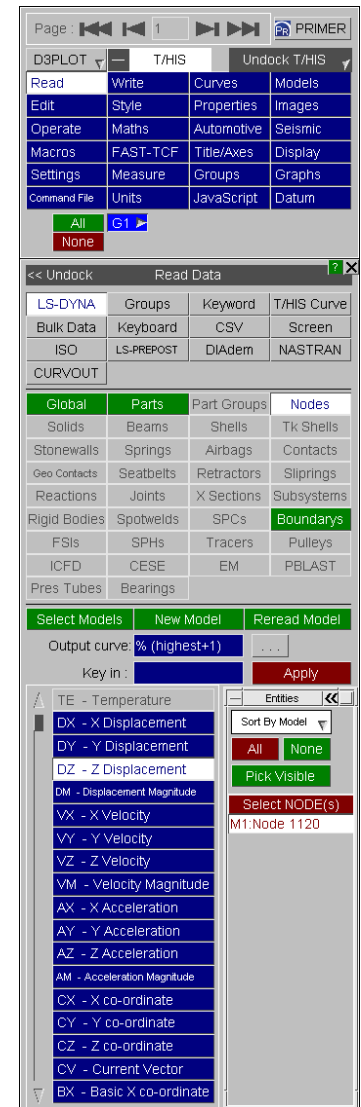
Viewing Commands

These are different to PRIMER and D3PLOT

- Click on **T/HIS** button via the Main Menu Option - this will open a T/HIS session within D3PLOT.
- Click **"Nodes"**.
- Select **'DZ – Z Displacement'**.
- Click **'Apply'** to plot data graph.
- Note: Left clicking a line on the graph blanks it. Press middle mouse button to unblank.
- Further graph and curve options can be found under the **'T/HIS'** main menu panel -> **'Curves'**.



While D3PLOT allows visual representation of this result and interaction with it, T/HIS allows for plotting and manipulation of time histories. All data for post processing must be requested in the database cards when setting up the run.



<< Undock

Read Data

LS-DYNA

Bulk Data

ISO

CURVOUT

Groups

Keyboard

LS-PREPOST

Keyword

CSV

DIAdem

T/HIS Curve

Screen

NASTRAN

Global	Parts	Part Groups	Nodes
Solids	Beams	Shells	Tk Shells
Stonewalls	Springs	Airbags	Contacts
Geo Contacts	Seatbelts	Retractors	Sliprings
Reactions	Joints	X Sections	Subsystems
Rigid Bodies	Spotwelds	SPCs	Boundaries
FSIs	SPHs	Tracers	Pulleys
ICFD	CESE	EM	PBLAST
Pres Tubes	Bearings		

Select Models

New Model

Reread Model

Output curve: % (highest+1)

...

Key in :

Apply

<.. Go Back

FX - BC motion X Force

FY - BC motion Y Force

FZ - BC motion Z Force

FR - Resultant BC motion forc

MX - BC motion X Moment

MY - BC motion Y Moment

MZ - BC motion Z Moment

MM - BC Moment Magnitude

EN - Energy from BC motion

Entities

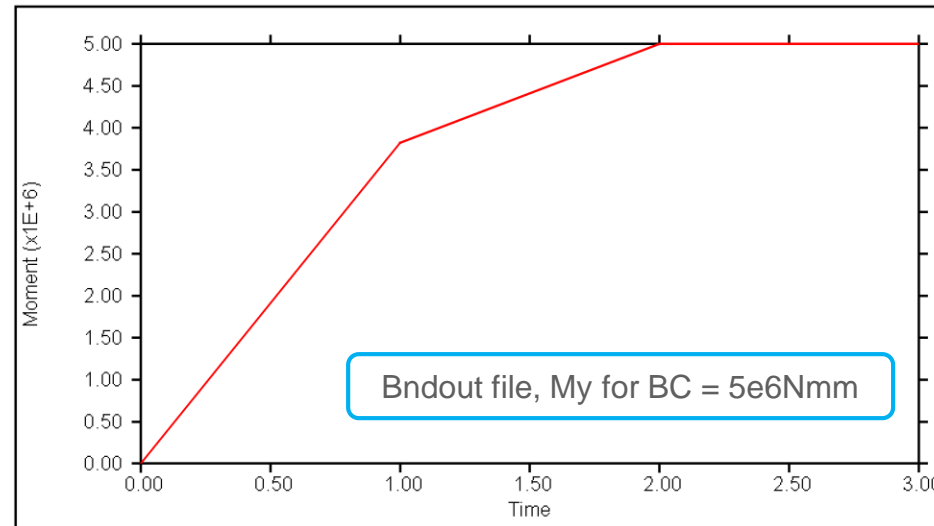
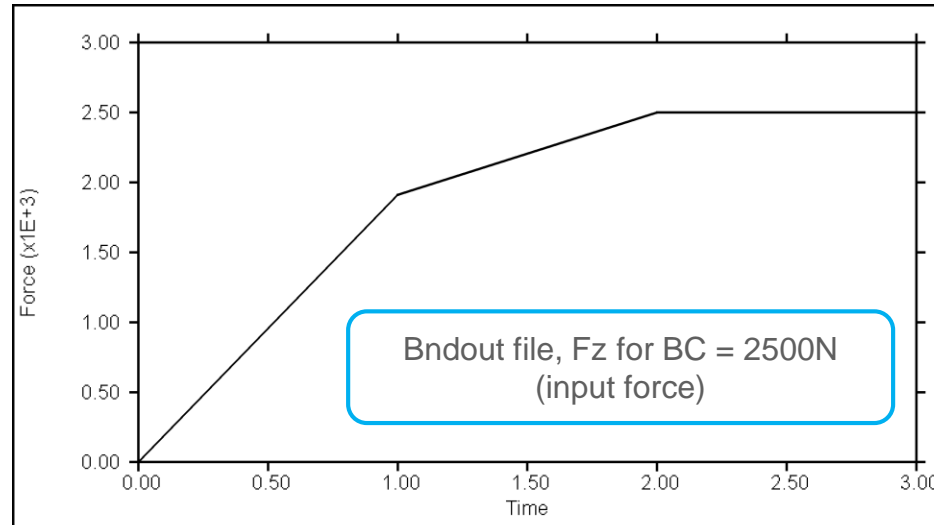
Sort By Model

All

None

Select RIGID BODY

M1:Rigid Body 140



**Check
reaction
forces and
moments**

Hand calculation: Tip Deflection

$$d = \frac{PL^3}{3EI} \left(1 + \frac{3f_s EI}{GAL^2} \right) \text{ where } f_s = \frac{A}{A_{web}}$$

$$E = 205000 \text{ N/mm}^2$$

$$P = 5000 \text{ N}$$

$$L = 2000 \text{ mm}$$

$$\nu = 0.3$$

$$\text{Density} = 7.85 \times 10^{-9} \text{ T/mm}^3$$

$$I_y = 46902000 \text{ mm}^4$$

$$\text{T/mm}^3$$

$$d_z = \frac{PL^3}{3EI_y} \left(1 + \frac{3f_s EI_y}{GAL^2} \right) = \frac{2.5 \cdot 2000^3}{3 \cdot 20500 \cdot 4690200} \left(1 + \frac{3 \cdot (1632.8/477.6) \cdot 20500 \cdot 4690200}{7.88 \times 10^4 \cdot 1632.8 \cdot 2000^2} \right) = 13.9 \cdot (1 + 0.019) = 7.07 \text{ mm}$$

$$G = \frac{E}{2(1 + \nu)}$$

Hand calculation: Max Bending Moment

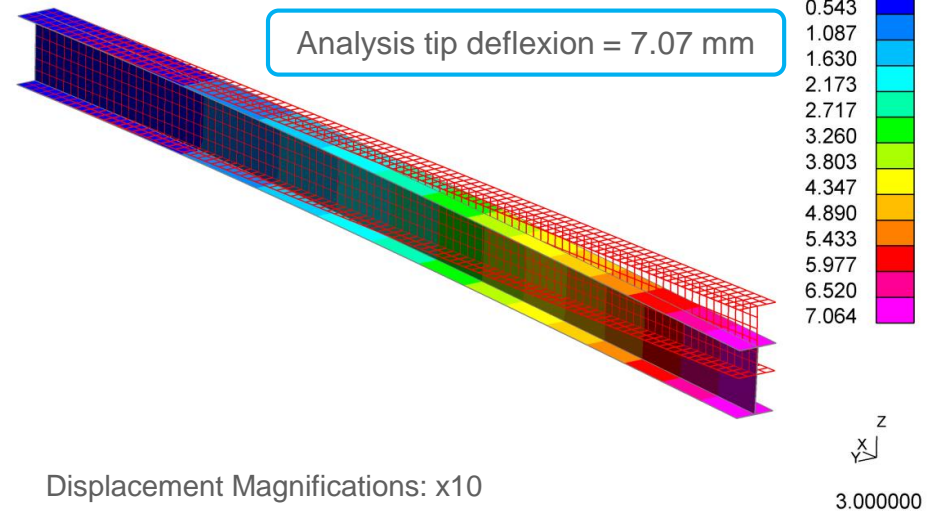
$$M_y = P \cdot L$$

$$M_y = P \cdot L = 2500 \cdot 2000 = 5 \times 10^6 \text{ Nmm}$$

Bndout file, My for BC = 5e6Nmm

Bndout file, Fz for BC = 2500N (input force)

D3PLOT:





www.arup.com/dyna

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